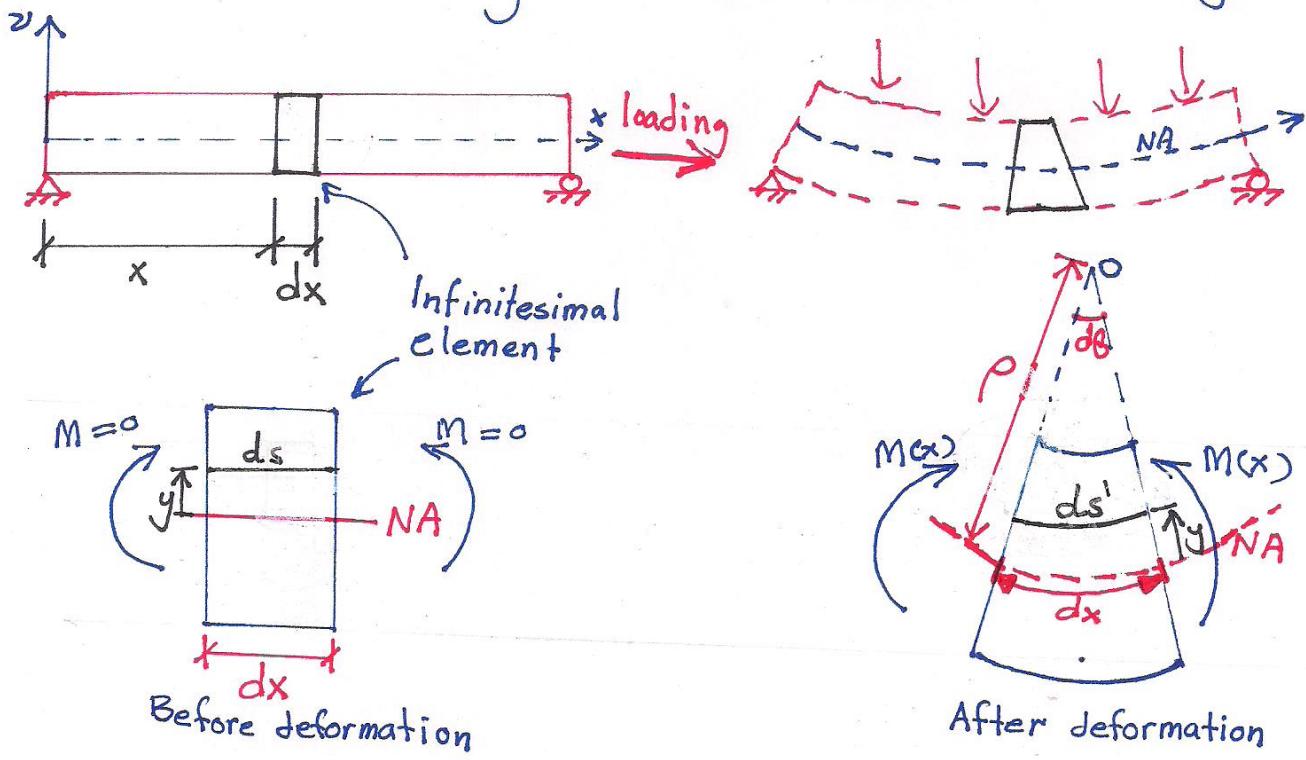


Elastic beam theory and double integration method

Consider the following beam, before and after loading:



$$\epsilon_x(y) = \frac{ds' - ds}{ds} = \frac{(R-y)dx - xdy}{Rdx} = -\frac{y}{R}$$

$\frac{1}{R} = -\frac{\epsilon_{xx}(y)}{y}$

"The infinitesimal element is subjected to normal strain (ϵ)"

For linear-elastic behavior
and Homogenous beam:

$$\epsilon_x = \frac{\sigma_x}{E} = \frac{-My}{I_z} = -\frac{My}{E \cdot I_z}$$

$$\frac{1}{R} = -\frac{(-My/E \cdot I_z)}{y}$$

$$\sigma_x = E \epsilon_x \text{ "Hooke's Law"}$$

$$\sigma_x = -\frac{My}{I_z} \text{ "Flexure formula"}$$

$$\boxed{\frac{1}{R} = -\frac{M}{EI_z}}$$

Curvature

Point "O": is the center of curvature

ρ : is the radius of curvature at the point of the elastic curve

$\frac{1}{\rho}$ = curvature : is the amount by which a curve deviates from being a straight line.

E = material's modulus of elasticity or Young's modulus

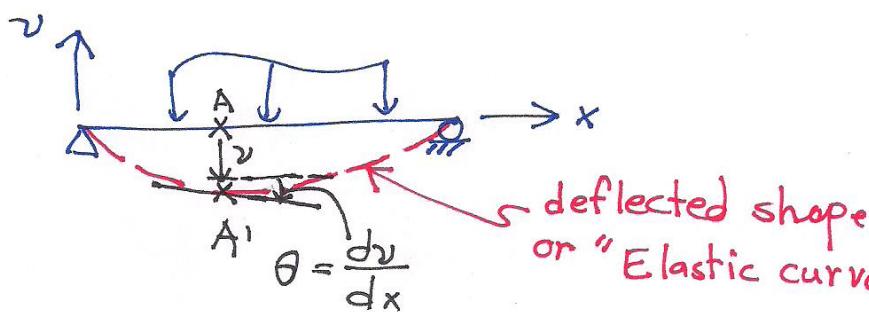
I = The beam's moment of inertia about the neutral axis (NA).

$$\frac{1}{\rho} = \frac{d^2v/dx^2}{2\sqrt{(1+(dv/dx)^2)^3}} = \frac{M}{EI}$$

nonlinear second-order differential equation gives exact shape of the elastic curve assuming that beam deflections occur only due to "bending"
→ Shear deformation is ignored

v = vertical deflection of the beam

$\theta = \frac{dv}{dx}$ = slope of the tangent line to the deformed beam (elastic curve)



or "Elastic curve" = deflection curve of the longitudinal axis that passes through the centroid of each cross-sectional area of the beam.

For linear-elastic behavior → deformation is very small

$$\theta = \frac{dv}{dx} \ll 1 \quad \theta^2 = \left(\frac{dv}{dx}\right)^2 \approx 0$$

$$\boxed{\frac{1}{\rho} = \frac{d^2v}{dx^2}}$$

This makes life much easier.

$$\frac{d^2 v}{dx^2} = \frac{M}{EI}$$

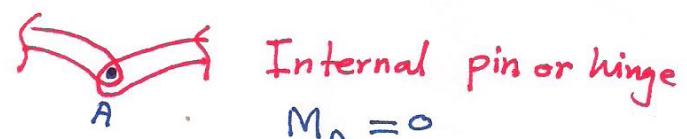
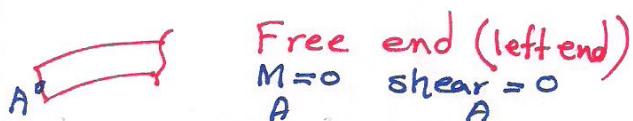
Second order ordinary differential equation.

$d^2 v / dx^2 = M/EI$ = curvature = γ_p = rate of change in slope
 M = internal moment in the beam at the point
 EI = flexural rigidity and it is always a positive quantity.

Steps to Calculate deflection by double integration method:

- Calculate support reactions, if necessary.
- Write the expression of the internal bending moment $M(x) = ?$
- Write the boundary and continuity conditions
- $v = \int \int \frac{M}{EI} dx$ $\theta = \frac{dv}{dx} = \int \frac{m}{EI} dx$
- determine the constants of integration using the conditions specified in part "c".

Boundary Conditions

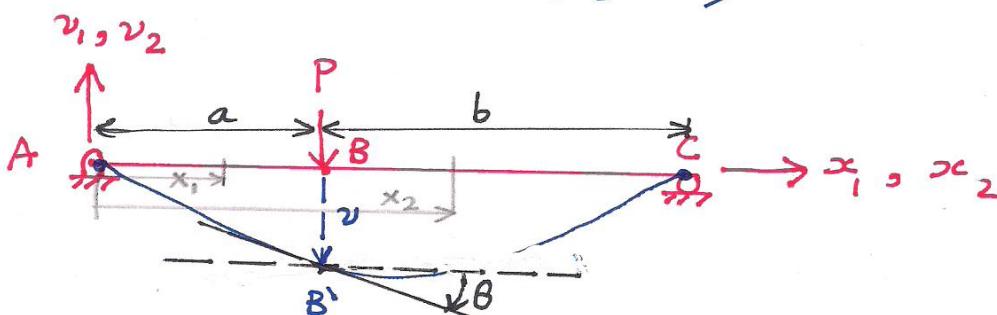


Continuity Conditions

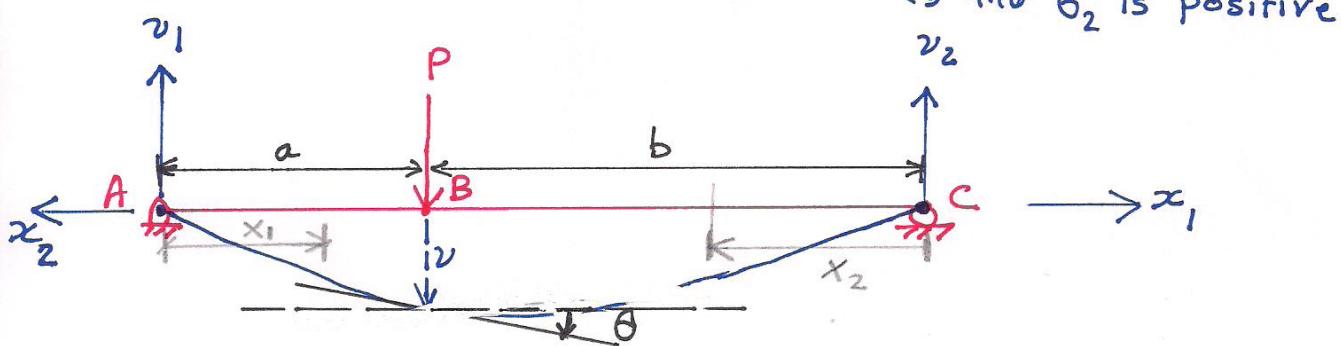
For the beam shown, two x -coordinates are chosen with origins at A. Each is valid only within the regions $0 \leq x_1 \leq a$ and $a \leq x_2 \leq (a+b)$. Once the functions for the slope and deflection are obtained, they must give the same values for the slope and deflection at point B so the elastic curve is physically continuous.

$$\theta_1(x_1=a) = \theta_2(x_2=a)$$

$$v_1(x_1=a) = v_2(x_2=a)$$

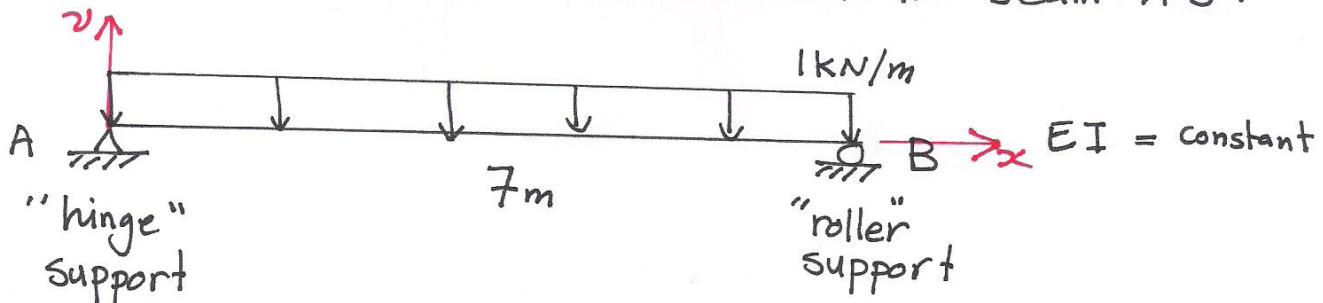


If instead the elastic curve is expressed in terms of the coordinates $0 \leq x_1 \leq a$ and $0 \leq x_2 \leq b$, as shown below, then the continuity of slope and deflection at B requires $\theta_1(x_1=a) = -\theta_2(x_2=b)$ and $v_1(x_1=a) = v_2(x_2=b)$. In this particular case, a negative sign is necessary to match the slopes at B since x_1 extends positive to the right, whereas x_2 extends positive to the left. Consequently, θ_1 is positive counter-clockwise, and θ_2 is positive clockwise.



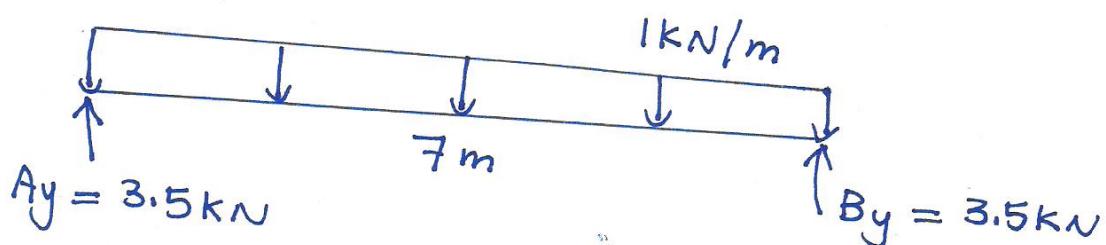
Example

Use the double integration method to determine the location and magnitude of maximum deflection for beam A-B.



Solution

a) Determine the reactions at A and B:



b) Write the internal moment as a function of x :

The diagram shows the beam with a coordinate x starting from the hinge support A. The moment M is shown as a single function. The equation for the moment is derived from the sum of moments at the hinge support A:

$$\text{At } \sum M_A = 0 : +M + (1)(x)\left(\frac{x}{2}\right) - 3.5x = 0$$

$$M = -\frac{x^2}{2} + 3.5x$$

$$0 \leq x \leq 7\text{ m}$$

Note: "M" is represented by single function

\Rightarrow Each will have single function

\Rightarrow No continuity conditions

$$c) \frac{d^2 v}{dx^2} = \frac{M}{EI}$$

$$EI \cdot \frac{d^2 v}{dx^2} = M \rightarrow EI \frac{d^2 v}{dx^2} = -\frac{x^2}{2} + 3.5x$$

$$EI \int \frac{d^2 v}{dx^2} dx = \int \left(-\frac{x^2}{2} + 3.5x \right) dx \rightarrow EI \frac{dv}{dx} = -\frac{x^3}{6} + \frac{3.5}{2} x^2 + C_1$$

slope

$$EI \int \frac{dv}{dx} dx = \int \left(-\frac{x^3}{6} + \frac{3.5}{2} x^2 + C_1 \right) dx$$

$$EI v = -\frac{x^4}{24} + \frac{3.5}{6} x^3 + C_1 x + C_2$$

$$C_1 = ? \quad \text{and} \quad C_2 = ?$$

d) boundary conditions:

Hinge support at A: deflection = 0 $v_A = 0$ at $x=0$

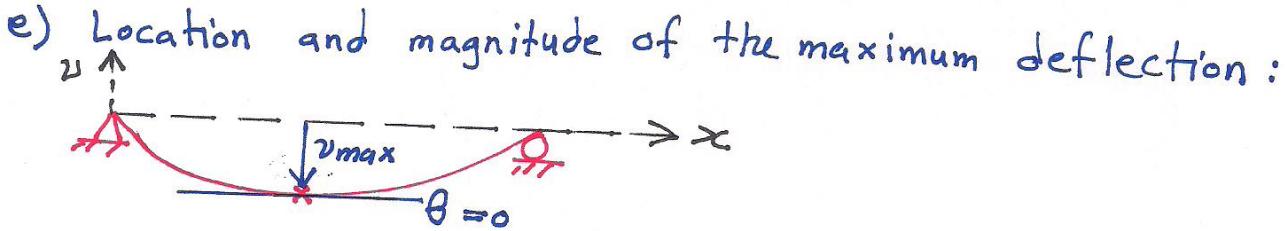
Roller support at B: $v_B = 0$ at $x = 7m$

At $x=0$: $EI(0) = 0 + 0 + 0 + C_2 \rightarrow C_2 = 0$

At $x=7m$: $EI(0) = -\frac{(7)^4}{24} + \frac{3.5(7)^3}{6} + C_1(7) \rightarrow C_1 = -\frac{343}{24}$

$$\frac{dv}{dx} = \frac{1}{EI} \left[-\frac{x^3}{6} + \frac{3.5}{2} x^2 - \frac{343}{24} \right] \quad \text{"slope" radian}$$

$$v = \frac{1}{EI} \left[\frac{-x^4}{24} + \frac{3.5}{6} x^3 - \frac{343}{24} x \right] \quad \text{"deflection" meter}$$



→ The slope is zero at the maximum deflection

$$\frac{dv}{dx} = 0 ; \quad -\frac{x^3}{6} + \frac{3.5}{2}x^2 - \frac{343}{24} = 0$$

$$x = 3.5 \text{ m}$$

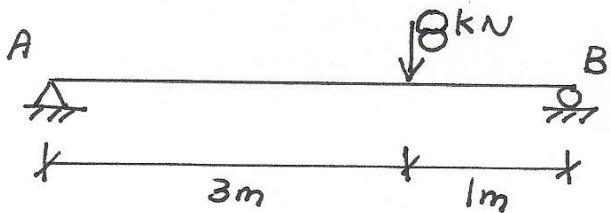
$$v_{max} = v \Big|_{x=3.5 \text{ m}} = \frac{1}{EI} \left[-\frac{3.5^4}{24} + \frac{3.5(3.5)^3}{6} - \frac{343}{24}(3.5) \right]$$

$$= \frac{-12005}{384EI} \text{ m} \quad (\downarrow)$$

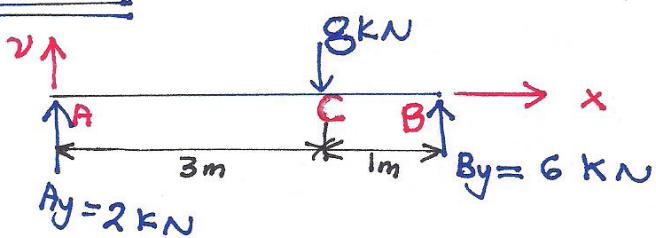
$$v_{max} = \frac{5wL^4}{384EI}$$

Example

For the beam shown, calculate the maximum deflection



Solution



Part AC

$$\begin{aligned} & \text{At } A: \uparrow v_1, \downarrow M_1 \\ & \text{At } C: \uparrow v_1, \downarrow M_1, \leftarrow 2kN \\ & \text{At } B: \uparrow v_1, \downarrow M_1, \uparrow By = 6kN \\ & \sum M_A = 0: +M_1 - 2x_1 = 0 \\ & M_1 = 2x_1 \\ & 0 \leq x_1 \leq 3 \end{aligned}$$

$$\begin{aligned} & EI \frac{d^2v_1}{dx_1^2} = 2x_1 \rightarrow EI \frac{d^2v_1}{dx_1^2} = x_1^2 + c_1 \\ & \int \rightarrow EI v_1 = \frac{x_1^3}{3} + c_1 x_1 + c_2 \end{aligned}$$

$$\text{Boundary Conditions: } v_1 \Big|_{x_1=0} = 0$$

$$\text{Continuity Conditions at } C: v_1 \Big|_{x_1=3m} = v_2 \Big|_{x_2=3m}$$

$$\frac{d^2v_1}{dx_1^2} \Big|_{x_1=3m} = \frac{d^2v_2}{dx_2^2} \Big|_{x_2=3m}$$

Part CB

$$\begin{aligned} & \text{At } C: \uparrow v_2, \downarrow M_2 \\ & \text{At } B: \uparrow v_2, \downarrow M_2, \uparrow By = 6kN \\ & \sum M_C = 0: +M_2 + 8(x_2 - 3) - 2x_2 = 0 \\ & M_2 = -6x_2 + 24 \\ & 3m \leq x_2 \leq 4m \end{aligned}$$

$$EI \frac{d^2v_2}{dx_2^2} = -6x_2 + 24$$

$$EI \frac{d^2v_2}{dx_2^2} = -3x_2^2 + 24x_2 + c_3$$

$$EI v_2 = -\frac{x_2^3}{3} + 12x_2^2 + c_3 x_2 + c_4$$

$$v_2 \Big|_{x_2=4m} = 0$$

$$\left. v_1 \right|_{x_1=0} = 0 \Rightarrow 0 = 0 + 0 + c_2 \Rightarrow c_2 = 0$$

$$\left. v_2 \right|_{x_2=4} = 0 \Rightarrow 0 = -(4)^3 + 12(4)^2 + c_3(4) + c_4 \quad \text{--- (1)}$$

$$\left. v_1 \right|_{x_1=3m} = \left. v_2 \right|_{x_2=3} \Rightarrow \frac{(3)^3}{3} + c_1(3) = -(3)^3 + 12(3)^2 + c_3(3) + c_4 \quad \text{--- (2)}$$

$$\left. \frac{dv_1}{dx_1} \right|_{x_1=3m} = \left. \frac{dv_2}{dx_2} \right|_{x_2=3m} \Rightarrow (3)^2 + c_1 = -3(3)^2 + 24(3) + c_3 \quad \text{--- (3)}$$

Solve equations 1, 2, and 3 for the constants c_1, c_3, c_4 :

$$c_1 = -5, \quad c_3 = -41, \quad c_4 = 36$$

$$\frac{dv_1}{dx_1} = \frac{1}{EI} \left[x_1^2 - 5 \right] \quad 0 \leq x_1 \leq 3m$$

$$v_1 = \frac{1}{EI} \left[\frac{x_1^3}{3} + -5x_1 \right]$$

$$\frac{dv_2}{dx_2} = \frac{1}{EI} \left[-3x_2^2 + 24x_2 - 41 \right] \quad 3m \leq x_2 \leq 4m$$

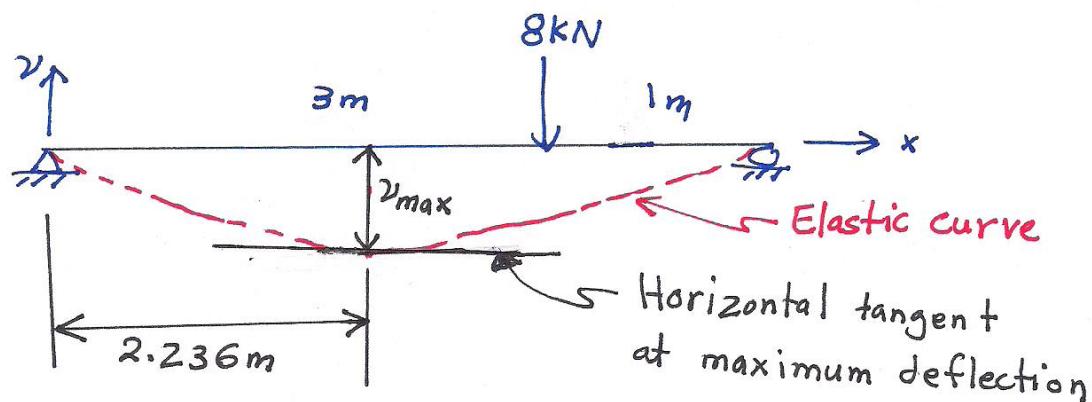
$$v_2 = \frac{1}{EI} \left[-\frac{x_2^3}{3} + 12x_2^2 - 41x_2 + 36 \right]$$

Assume the maximum deflection is within part AC:

$$\frac{d^2v_1}{dx_1^2} = 0 : \quad x_1^2 - 5 = 0 \rightarrow x_1 = \sqrt{5} = 2.236m < 3m$$

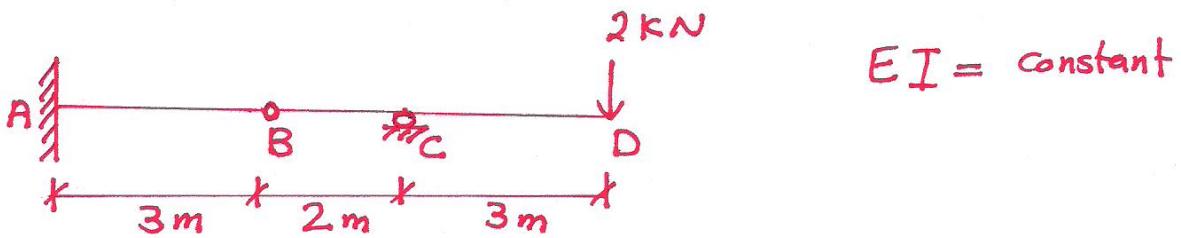
$$v_{\max} = v_1 \Big|_{x_1 = 2.236m} = \frac{1}{EI} \left[\frac{(2.236)^3}{3} - 5(2.236) \right]$$

$$= \frac{-7.453}{EI} m (\downarrow)$$



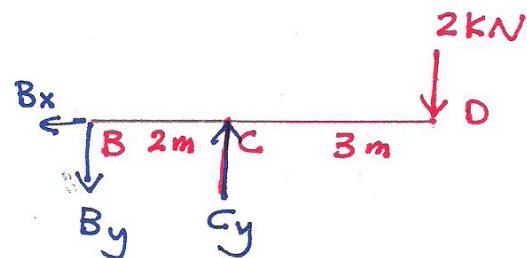
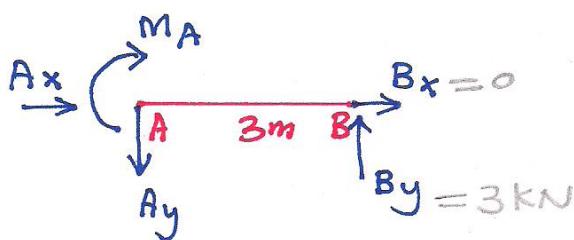
Example

Beam ABCD shown has a fixed support at A, an internal hinge at B, a roller support at C, and a free end at D. Determine the deflection and slope at points B and D.



Solution

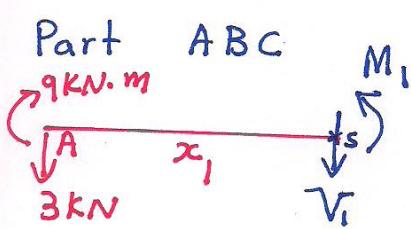
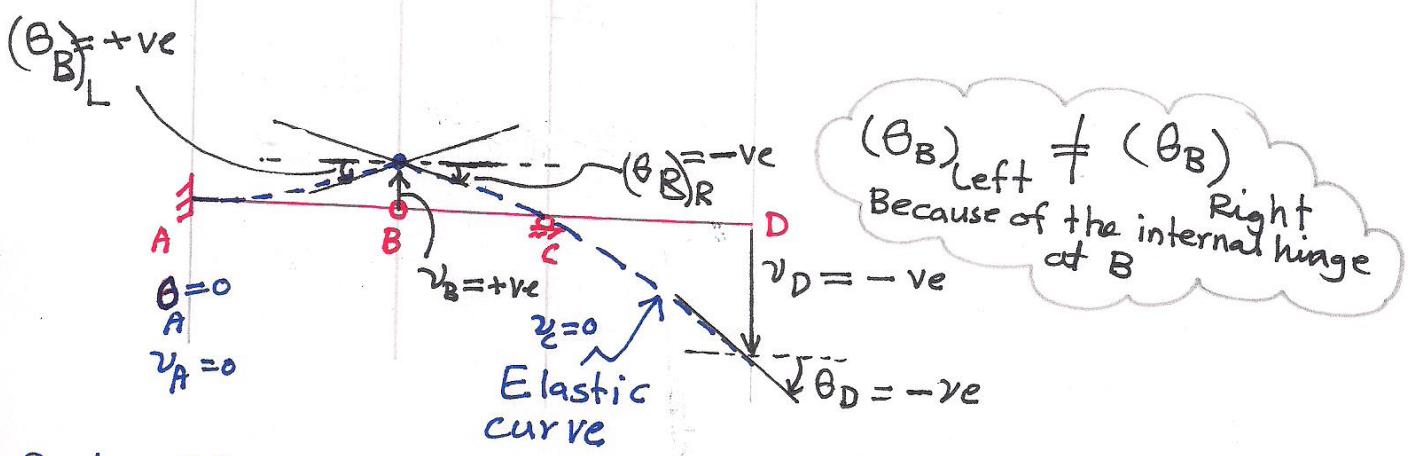
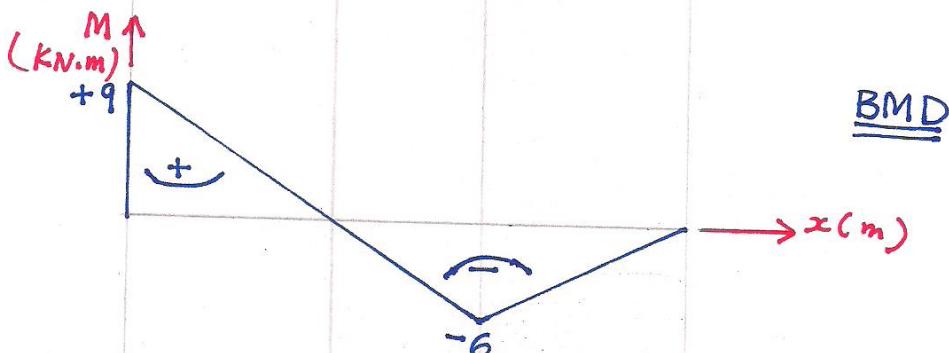
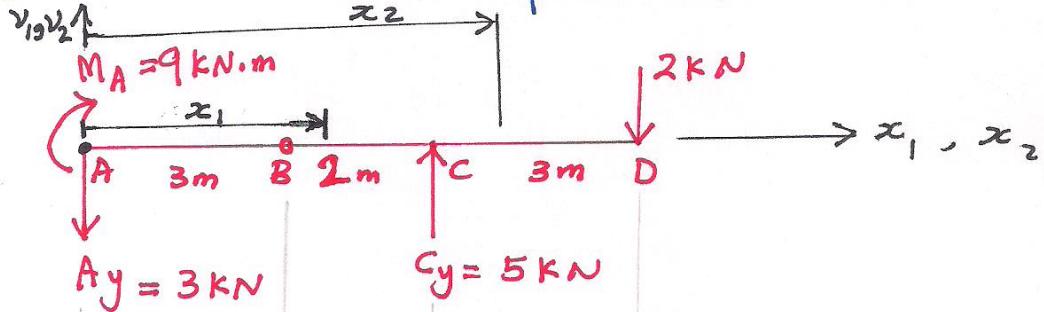
Determine the reactions at A and C



$$\begin{aligned} \sum M_B &= 0: -M_A + 3(3) = 0 \\ M_A &= 9 \text{ kN.m} \end{aligned}$$

$$\begin{aligned} \sum F_y &= 0: -B_y + 5 - 2 = 0 \\ B_y &= 3 \text{ kN} (\downarrow) \end{aligned}$$

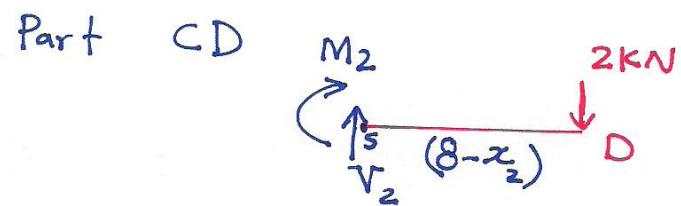
Write the moment expression:



$$(+ \sum M_s = 0 : +M_1 + 3x_1 - 9 = 0)$$

$$M_1 = 9 - 3x_1$$

$$0 \leq x_1 \leq 5 \text{ m}$$



$$(+ \sum M_s = 0 : -M_2 - 2(8 - x_2) = 0)$$

$$M_2 = -16 + 2x_2$$

$$5 \text{ m} \leq x_2 \leq 8 \text{ m}$$

Part A B:

$$EI \frac{d^2 v_{AB}}{dx_1^2} = 9 - 3x_1 \quad \int \quad EI \frac{dv_{AB}}{dx_1} = 9x_1 - \frac{3}{2}x_1^2 + C_1$$

$\downarrow \int$

$$EI v_{AB} = \frac{9}{2}x_1^2 - \frac{1}{2}x_1^3 + C_1 x_1 + C_2$$

$$\left. v_{AB} \right|_{x_1=0} = 0 : 0 = 0 - 0 + 0 + C_2 \rightarrow C_2$$

$$\left. \frac{dv_{AB}}{dx_1} \right|_{x_1=0} = 0 : 0 = 0 - 0 + C_1 \rightarrow C_1 = 0$$

$$\frac{dv_{AB}}{dx_1} = \frac{1}{EI} \left[9x_1 - \frac{3}{2}x_1^2 \right] \quad 0 \leq x_1 \leq \underbrace{3 \text{ m}}$$

$$v_{AB} = \frac{1}{EI} \left[\frac{9}{2}x_1^2 - \frac{1}{2}x_1^3 \right]$$

not to 5 m
 \Rightarrow There is
 a discontinuity
 at B because
 of the internal
 hinge.

Part BC

$$3 \text{ m} \leq x_1 \leq 5 \text{ m}$$

$$EI \frac{d^2 v_{BC}}{dx_1^2} = 9 - 3x_1 \quad \int \quad EI \frac{dv_{BC}}{dx_1} = 9x_1 - \frac{3}{2}x_1^2 + C_3$$

$\downarrow \int$

$$\left. v_{BC} \right|_{x_1=5} = 0 : 0 = \frac{9}{2}(5)^2 - \frac{(5)^3}{2} + C_3(5) + C_4$$

$$EI v_{BC} = \frac{9}{2}x_1^2 - \frac{x_1^3}{2} + C_3 x_1 + C_4$$

$$5C_3 + C_4 = -50 \quad \text{---} ①$$

$$\left. v_{AB} \right|_{x_1 = 3m} = \left. v_{BC} \right|_{x_1 = 3m}$$

$$\frac{1}{EI} \left[\frac{9}{2}(3)^2 - \frac{1}{2}(3)^3 \right] = \frac{1}{EI} \left[\frac{9}{2}(3)^2 - \frac{(3)^3}{2} + C_3(3) + C_4 \right]$$

$$3C_3 + C_4 = 0 \quad \textcircled{2}$$

From equations ① and ②: $C_3 = -25$

$$C_4 = 75$$

$$\frac{d^2v_{BC}}{dx_1^2} = \frac{1}{EI} \left[9x_1 - \frac{3}{2}x_1^2 - 25 \right] \quad 3m \leq x_1 \leq 5m$$

$$v_{BC} = \frac{1}{EI} \left[\frac{9}{2}x_1^2 - \frac{x_1^3}{2} - 25x_1 + 75 \right]$$

Part CD $5m \leq x_2 \leq 8m$

$$EI \frac{d^2v_{CD}}{dx_2^2} = -16 + 2x_2 \quad \xrightarrow{\int} EI \frac{dv_{CD}}{dx_2} = -16x_2 + x_2^2 + C_5$$

$$\downarrow \\ EI v_{CD} = -8x_2^2 + \frac{x_2^3}{3} + C_5 x_2 + C_6$$

$$\left. v_{CD} \right|_{x_2=5m} = 0 \quad ; \quad 0 = -8(5)^2 + \frac{(5)^3}{3} + 5C_5 + C_6$$

$$5C_5 + C_6 = \frac{475}{3} \quad \textcircled{3}$$

$$\left. \frac{d^2 v_{BC}}{dx_1} \right|_{x_1=5m} = \left. \frac{d^2 v_{CD}}{dx_2} \right|_{x_2=5m}$$

$$\frac{1}{EI} \left[9(5) - \frac{3(5)^2}{2} - 25 \right] = \frac{1}{EI} \left[-16(5) + (5)^2 + C_5 \right]$$

$$C_5 = 37.5$$

Use equation ③ to find $C_6: C_6 = -\frac{175}{6}$

$$\frac{d^2 v_{CD}}{dx_2} = \frac{1}{EI} \left[-16x_2 + x_2^2 + 37.5 \right] \quad 5m \leq x_2 \leq 8m$$

$$v_{CD} = \frac{1}{EI} \left[-8x_2^2 + \frac{x_2^3}{3} + 37.5x_2 - \frac{175}{6} \right]$$

$$(B_B)_L = \left. \frac{d^2 v_{AB}}{dx_1} \right|_{x_1=3m} = \frac{1}{EI} \left(9(3) - \frac{3}{2}(3)^2 \right) = \frac{+27}{2EI} \text{ rad} \quad \cancel{\downarrow \theta_L}$$

$$(B_B)_R = \left. \frac{d^2 v_{BC}}{dx_1} \right|_{x_1=3m} = \frac{1}{EI} \left[9(3) - \frac{3}{2}(3)^2 - 25 \right] = \frac{-23}{2EI} \text{ rad} \quad \cancel{\downarrow \theta_R}$$

$$v_B = \left. v_{AB} \right|_{x_1=3m} = \left. v_{BC} \right|_{x_1=3m} = \frac{1}{EI} \left[\frac{9}{2}(3)^2 - \frac{1}{2}(3)^3 \right] = \frac{+27}{EI} \text{ m} \quad (\uparrow)$$

$$\text{or } \frac{1}{EI} \left[\frac{9}{2}(3)^2 - \frac{1}{2}(3)^3 - 25(3) + 75 \right] = \frac{+27}{EI} \text{ m}$$

$$\theta_D = \left. \frac{d^2 v_{CD}}{dx_2} \right|_{x_2=8m} = \frac{1}{EI} \left[-16(8) + (8)^2 + 37.5 \right] = \frac{-53}{2EI} \text{ rad} \quad \cancel{\downarrow \theta_D}$$

$$v_D = \left. v_{CD} \right|_{x_2=8m} = \frac{1}{EI} \left[-8(8)^2 + \frac{(8)^3}{3} + 37.5(8) - \frac{175}{6} \right] = -\frac{141}{2EI} \text{ m} \quad (\downarrow)$$

The maximum deflection is at point D which the free end