

# Chapter 8 Internal flow

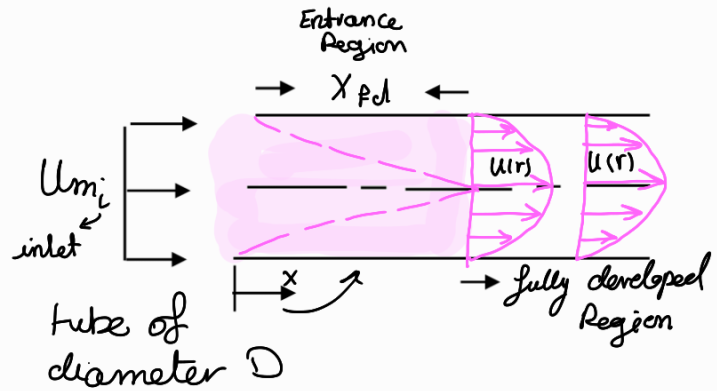
For flowing fluid

$$Re_D = \frac{U_m D}{\nu}$$

$\dot{m} = \rho V A$   
 Velocity =  $U_m$   
 $\mu = \rho \nu$

For tube flow:

- if  $Re_D < 2300$ : laminar
- if  $Re_D \geq 2300$ : turbulent



$x_{f,d}$ : distance to fully developed Region

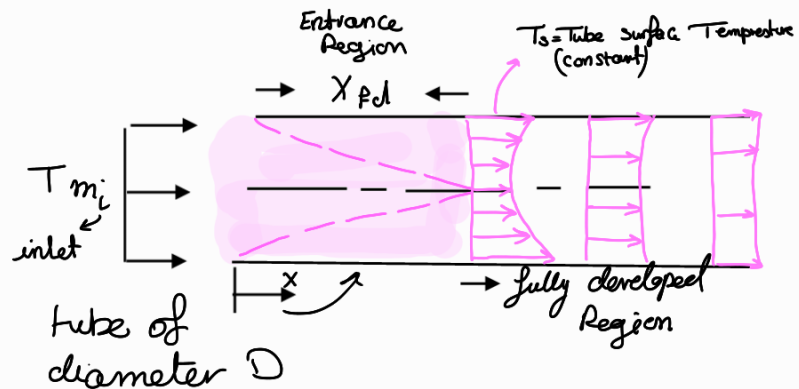
Note,  $Re_D = \frac{1 \dot{m}}{\pi D \mu}$

For temperature flow

$$\left(\frac{x_{f,d}}{D}\right)_{lam} = 0.05 Re_D Pr$$

$$\left(\frac{x_{f,d}}{D}\right)_{Turb} = 10$$

Always calculate it

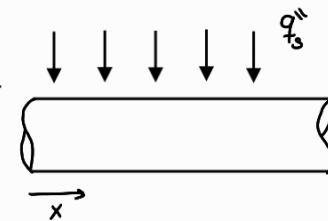
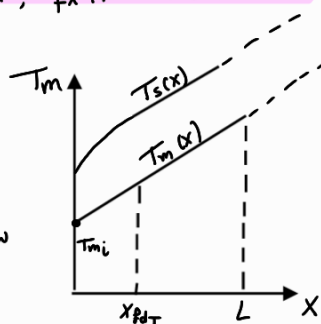


Constant surface Heat flux,  $q_s''$  (for Turb or lam)

$$T_m(x) = T_{m,i} + \frac{q_s'' P}{\dot{m} c_p} x$$

when  $x=0$ ,  $T_m = T_{m,i}$

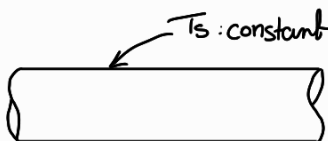
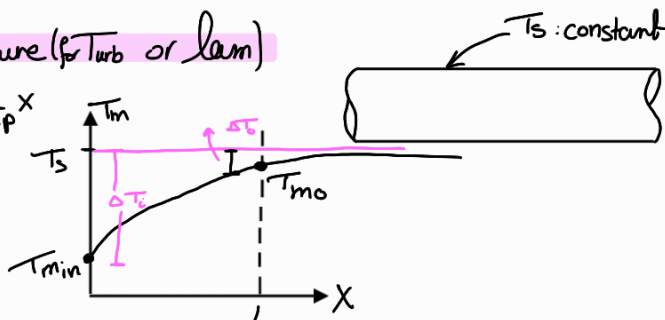
$$T_{s,0} = T_{m,0} + \frac{q_s''}{h} \rightarrow \text{for the flow}$$



Constant surface temperature (for Turb or lam)

$$\frac{T_s - T_m(x)}{T_s - T_{m,i}} = e^{-\frac{Ph}{\dot{m} c_p} x}$$

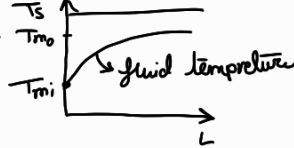
$$P = \pi D$$



$q = \bar{h} A_s \Delta T_{lm}$  where  $\Delta T_{lm} = \frac{\Delta T_o - \Delta T_i}{\ln\left(\frac{\Delta T_o}{\Delta T_i}\right)}$  log mean temperature difference

Also

$q = \dot{m} C_p (T_{mo} - T_{mi}) \rightarrow q_s \text{ const or } T_s \text{ const}$



Arithmetic mean temperature difference:

$$\Delta T_{am} = \frac{\Delta T_o + \Delta T_i}{2} = \frac{(T_s - T_o) + (T_s - T_i)}{2}$$

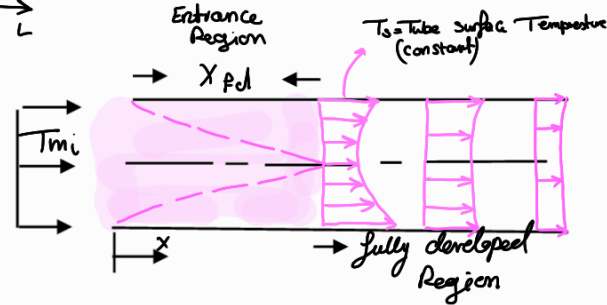
To find  $\bar{h}$  for tubes (laminar flow)

- For  $q_s''$  constant

At  $T_m \rightarrow Nu_D = \frac{hD}{K} = 1.34 \rightarrow$  This is for  $\left\{ \begin{array}{l} \text{fully developed conditions} \\ \text{fully developed velocity} \\ \text{fully developed temperature} \end{array} \right.$

- For  $T_s$  constant

At  $T_m \rightarrow Nu_D = \frac{hD}{K} = 3.66$



- For entrance Region (Thermal entry)

$Nu_D = \frac{hD}{K} = 3.66 + \frac{0.0668 Gr_{zD}}{1 + 0.01(Gr_{zD})^{2/3}}$

where  $Gr_{zD} = \left(\frac{D}{x}\right) Re_D Pr$  Graetz number

$T_s = \text{constant}$

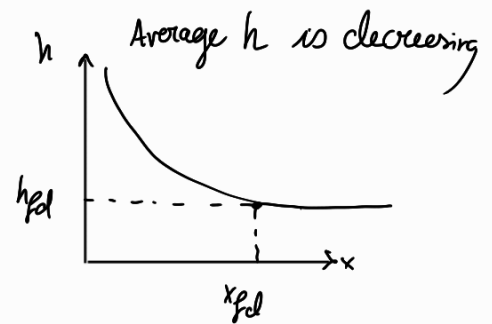
Properties of  $\bar{T}_m = \frac{T_{mi} + T_{mo}}{2}$

Combined entry length

$$\bar{Nu}_D = \frac{3.66}{\tanh\left[2.261 Gr_{zD}^{-1/3} + 1.7 Gr_{zD}^{-2/3}\right]} + 0.0499 Gr_{zD} \tanh(Gr_{zD})$$

$$\tanh(2.432 Pr^{1/4} Gr_{zD}^{-1/6})$$

Note



To find  $\bar{h}$  for tubes (Turbulent flow)  $\rightarrow$  smooth (fully developed)

$\bar{Nu}_D = 0.023 Re_D^{4/5} Pr^n$

$n = 0.4$  For heating  $T_s > T_m$   
 $n = 0.3$  For cooling  $T_s < T_m$

- Prop at  $\bar{T}_m$

- Used for  $T_s \text{ const}$  or  $q_s \text{ constant}$

If the tube is rough

$$Nu_D = \frac{(f/8)(Re_D - 1000)Pr}{1 + 12.7(f/8)^{1/2}(Pr^{1/3} - 1)}$$

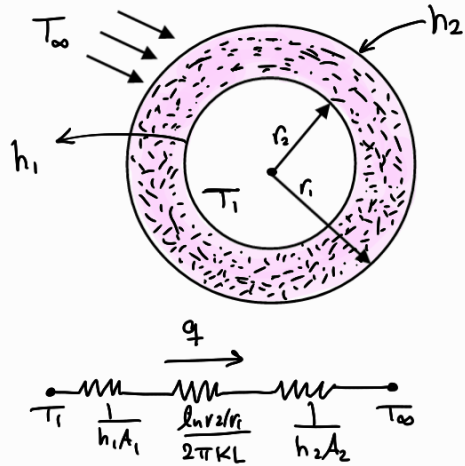
$f$ : from moody chart

valid for  $3000 \leq Re_D \leq 5 \times 10^6$

Overall heat transfer coefficient

$$\frac{\Delta T_o}{\Delta T_i} = \frac{T_{\infty o} - T_{m,o}}{T_{\infty i} - T_{m,i}} = e^{\left(\frac{-\bar{U} A_s}{\dot{m} c_p}\right)} \rightarrow A_s = \pi D L$$

$$\dot{q} = \bar{U} A_s \Delta T_{lm}$$



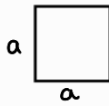
Noncircular Tubes

hydraulic diameter

$$D_h = \frac{4A_c}{P} \rightarrow \text{flow cross-sectional area}$$

$P$  → wetted perimeter  
used instead of  $D$

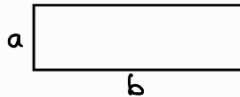
• For square section



$$D_h = a$$

• For rectangular section

$$D_h = \frac{2ab}{a+b}$$



• Parallel sheets  $b \gg a$

$$D_h = 2a$$

$$Re = \frac{\dot{m} D_h}{A_c \mu} = \frac{\dot{m} \left(\frac{2ab}{a+b}\right)}{ab \mu}$$



TABLE 8.1 Nusselt numbers and friction factors for fully developed laminar flow in tubes of differing cross section

| Cross Section | $\frac{b}{a}$ | $Nu_D = \frac{hD_h}{k}$ |                  | $fRe_{D_h}$ |
|---------------|---------------|-------------------------|------------------|-------------|
|               |               | (Uniform $q_s''$ )      | (Uniform $T_s$ ) |             |
| Circle        | —             | 4.36                    | 3.66             | 64          |
| Square        | 1.0           | 3.61                    | 2.98             | 57          |
| Rectangle     | 1.43          | 3.73                    | 3.08             | 59          |
| Rectangle     | 2.0           | 4.12                    | 3.39             | 62          |
| Rectangle     | 3.0           | 4.79                    | 3.96             | 69          |
| Rectangle     | 4.0           | 5.33                    | 4.44             | 73          |
| Rectangle     | 8.0           | 6.49                    | 5.60             | 82          |
| Heated        | $\infty$      | 8.23                    | 7.54             | 96          |
| Insulated     | $\infty$      | 5.39                    | 4.86             | 96          |
| Triangle      | —             | 3.11                    | 2.49             | 53          |

Used with permission from W. M. Kays and M. E. Crawford, *Convection Heat and Mass Transfer*, 3rd ed. McGraw-Hill, New York, 1993.

For cases not in the table, use  $D_h$  in previous correlations obtained

But  $P$ ,  $A$  are calculated for original shape

## Concentric tube annulus

For interior:

$$q''_i = h_i (T_{s,i} - T_{m,i})$$

Tables 8.2, 8.3

$$Nu_i = \frac{h_i D_h}{k}$$

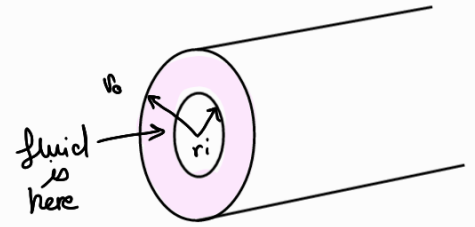
For outside

$$q''_o = h_o (T_{s,o} - T_{m,o})$$

$$Nu_o = \frac{h_o D_h}{k}$$

$$D_h = D_o - D_i \quad (\text{hydraulic diameter})$$

Generally  $q = \dot{m} c_p (T_{m,o} - T_{m,i})$



**TABLE 8.2** Nusselt number for fully developed laminar flow in a circular tube annulus with one surface insulated and the other at constant temperature

| $D_i/D_o$      | $Nu_i$ | $Nu_o$ | Comments                                |
|----------------|--------|--------|---|
| 0              | —      | 3.66   | See Equation 8.55                       |
| 0.05           | 17.46  | 4.06   |   |
| 0.10           | 11.56  | 4.11   |   |
| 0.25           | 7.37   | 4.23   |   |
| 0.50           | 5.74   | 4.43   |   |
| $\approx 1.00$ | 4.86   | 4.86   | See Table 8.1, $b/a \rightarrow \infty$ |

Note

$$q'' = \frac{q}{A} = \frac{q' L}{A}$$

$$A = 2\pi r L$$

**TABLE 8.3** Influence coefficients for fully developed laminar flow in a circular tube annulus with uniform heat flux maintained at both surfaces

| $D_i/D_o$ | $Nu_{ii}$ | $Nu_{oo}$          | $\theta_i^*$ | $\theta_o^*$ |
|-----------|-----------|--------------------|--------------|--------------|
| 0         | —         | 4.364 <sup>a</sup> | $\infty$     | 0            |
| 0.05      | 17.81     | 4.792              | 2.18         | 0.0294       |
| 0.10      | 11.91     | 4.834              | 1.383        | 0.0562       |
| 0.20      | 8.499     | 4.833              | 0.905        | 0.1041       |
| 0.40      | 6.583     | 4.979              | 0.603        | 0.1823       |
| 0.60      | 5.912     | 5.099              | 0.473        | 0.2455       |
| 0.80      | 5.58      | 5.24               | 0.401        | 0.299        |
| 1.00      | 5.385     | 5.385 <sup>b</sup> | 0.346        | 0.346        |

Used with permission from W. M. Kays and H. C. Perkins, in W. M. Rohsenow and J. P. Hartnett, Eds., *Handbook of Heat Transfer*, Chap. 7, McGraw-Hill, New York, 1972.

- If the tube is placed in a fluid

$$\frac{T_{\infty} - T_{m,o}}{T_{\infty} - T_{m,i}} = e^{-\left(\frac{\bar{U} A_s}{\dot{m} c_p}\right)}$$

$$q = \bar{U} A_s \frac{\Delta T_{lm}}{\log \text{ mean}}$$

